

OZONEREPLENISHMENTWITHGASTURBINEOPER ATION

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ABSTRACT:

The current problem of the day is pollution, especially ozone layer depletion. The depletion is accelerated by aircraft propulsion which releases the impurities directly in the affect zone. The depletion accounts for the damage of all celled organisms. But, at the same time aviation cannot be ruled out as a mode of transportation for its time saving character in this fast blue ball. The project is aimed at making the aviation safer for the planet i.e., to reduce the amount of pollutants released by the aviation thrusters and also to accelerate the ozone production. The project will gain importance as it reduces pollution and replenishes ozone as these will be the major concern for tomorrow. The ozone is a combination of three oxygen atoms. The depletion is caused by the reaction of a member of the halogen family with ozone and also by the NO_x released from the aircraft's thrust producers.

We are going to split the water into its molecular components, hydrogen and nascent oxygen. The hydrogen thus produced is let into the combustion chamber and allowed to combust along with the hydrocarbon fuel and then nascent oxygen is released to react free in the atmosphere. The water is split by electrolysis. The combustion of hydrogen in the combustion chamber reduces the flame temperature. On temperatures above 1500°C , the production of NO_x starts, the addition of hydrogen reduces these temperature peaks and thus reduces the formation of NO_x . Direct water addition instead of hydrogen to the combustion chamber, reduces the efficiency of the chamber. Then nascent oxygen left out reacts with the halogen family or with the other agents which breaks up the ozone and reduces the possibility of depletion. At better cases, the nascent oxygen may well react with the oxygen molecule to form ozone with sunlight as catalyst.

Keywords: Ozonereplenishment; pollutionreductioninengines; Gasturbineengines; Aircraftpollutionreduction.

INTRODUCTION:

Ozone is a compound of oxygen that contains three atoms instead of the two found in the oxygen gas that sustains life. It was discovered in 1839 by a Swiss chemist, Christian Friedrich Schönbein. In high concentration, ozone is a bluish green gas, with very strong oxidizing properties. It is a toxic, irritating gas, often encountered in surface air pollution episodes, when it can trigger asthma and irritate mucous membranes. Dry air consists of 78% nitrogen and 21% oxygen and there are normally trace amounts of other gases, principally argon, water and carbon dioxide, present. The concentration of ozone is usually only a few parts per million and even in the ozone layer it is only one part in 100,000.

Ozone is created in the upper stratosphere by the photo-dissociation of an oxygen molecule, which liberates a free oxygen atom and this can then combine with another oxygen molecule to create ozone. The dissociation of the oxygen molecule requires ultraviolet light of wavelength shorter than 240 nm. Ozone itself can be dissociated by light of wavelength shorter than 1100 nm. The free oxygen atom thus created quickly finds another oxygen molecule and the ozone is reformed with the net result of absorbing the solar radiation and putting the energy into the atmosphere as thermal energy. The process is very efficient and virtually all radiation between 200 and 310 nm is absorbed, despite the relatively low concentration of ozone. The main ozone

absorption bands in the ultraviolet are the Hartley (around 200–300 nm) and Huggins (around 300 – 350 nm), and there is the weak Chappuis band in the visible (440 – 740 nm). In the lower stratosphere, below about 30 km, ozone has a long lifetime, and the ozone mixing ratio can be used to trace atmospheric motions.

In the normal state of affairs the creation and dissociation processes run in balance and a typical value for the total amount of ozone in a vertical column of our atmosphere is around 300 Dobson Units (DU), or 300 milli-atmosphere-centimetres, which corresponds to a layer of ozone 3 mm thick at the Earth's surface. This 3 mm is in reality spread through the column, with the bulk of it lying between the tropopause, at 10 to 12 km altitude, and 40 km, with a maximum at around 17 to 25 km altitude depending on location. This is the ozone layer.

Over the last 50 years we have introduced chemicals into the atmosphere that are capable of destroying ozone through photochemical processes. Chloro-fluorocarbons (CFCs) are widely known, but there are also other ozone depleting substances such as halons (bromo-fluorocarbons) and methyl

bromide. In certain circumstances the chlorine or bromine from these substances can react with ozone to turn it back into oxygen. In most parts of the world the reactions are very slow and there is little damage to the ozone layer, however over the Antarctic a dramatic hole opens in the ozone layer every spring and fills in again by mid-summer. This is created by the unusual atmospheric conditions that exist during the Antarctic winter.

An international treaty, the Montreal Protocol, has been drawn up to control the release of ozone depleting chemicals into the atmosphere. This treaty is clearly working, and the amount of these chemicals in air near the surface is beginning to decline. The chemicals are however so stable that it will take a long time before they drop to the level that existed 50 years ago and it is likely that we will see an annual ozone hole over Antarctica for many decades to come.

Aircrafts and ozone depletion

Aircrafts which we are operating is one among the major causes for ozone layer depletion and global warming. The exhaust generated by the aircrafts contains pollutants which break up the ozone molecules and causes depletion. The exhaust is such toxic due to incomplete combustion of the fuel in the combustion chamber and also due to the high temperatures.

METHODOLOGY:

We are going to split the project into two halves, one regards with ozone formation and the other with the pollution reduction.

At first the water has to be splitted into its molecular components and then the diversion starts. We use a hydro separator for these separation process and then inject nascent oxygen into the atmosphere and the hydrogen into the combustor.

WATER SEPERATION

Water is a combination of hydrogen and oxygen. There are many ways of splitting the water molecule into its molecular constituents. Many technologies have been explored but it should be noted that as of 2007 "Thermal, thermochemical, biochemical and photochemical processes have so far not found industrial applications." Only high temperature electrolysis of alkaline solutions finds some applications.

The various other methods of separation are as follows,

- Electrolysis
- Thermolysis
- Photocatalyticwatersplitting
- Sulfur-iodinecycle
- Biohydrogenroutes
- Fermentativehydrogenproduction
- Enzymatichydrogengeneration
- Biocatalysedelectrolysis

4.1Methodof choice

Based on the ease of the process, timetaken to generate the product and energy inputrequiredtheprocessesarecompared.TheElectrolysisprocesswhichrequiresjustaminimumof1.5Vfortheseparationisselected.

The separator configuration is180 Litres/Hour

InputVoltage(V)	:AC220
WorkingMedium	:WaterOperationGasPressure (KPa):1.5Rated
OutputGas(L/h)	200
Weight(kg)	75

TheHydroseparatorisshown inFig4.1

COMUUSTOR:

combustionchambersandtypes

Therearethreetypesofcombustionchambersusedforcommercialaircraftpropulsion.Theyare

- ✓ Cantype (Fig5.1)
- ✓ Annulartype(Fig5.2)
- ✓ Canannulartype (Fig5.3)
-

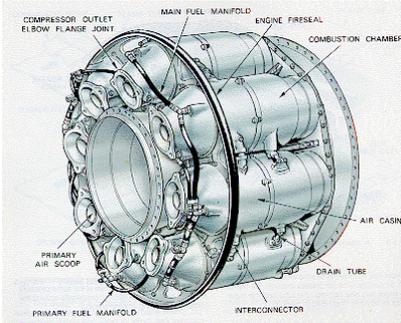


Fig5.1 Cantype combustor

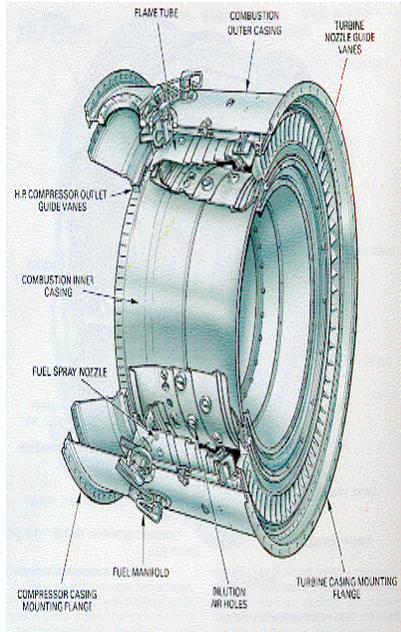


Fig4.1:Hydro Separator



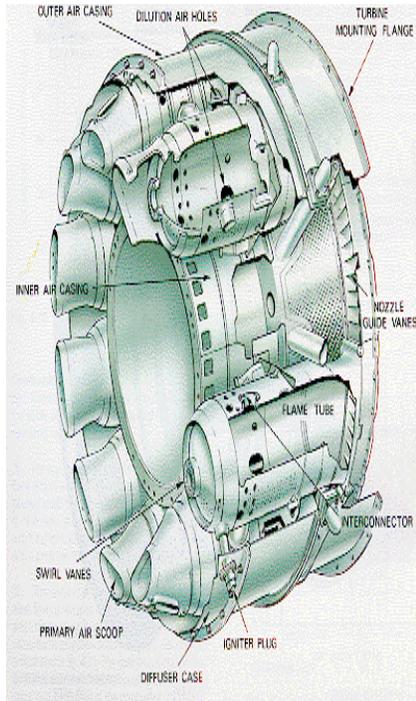


Fig5.3Canannular type combustor
Chamber of choice

RESULTS:

For our process can type is suitable. Why because, a single chamber can be used to test the result and can be integrated to get the result for the entire engine thus the amount of hydrogen required, fuel and air required are minimum.

DESCRIPTION

The fabricated combustion chamber setup is shown in Fig 6.1, the setup has a hydroseparator, a fuel tank and the combustor setup with lines for compressed air entry.



Fig6.1:CombustorSetup

Pollution reduction

The tests were carried out in such a way that at first the combustor was operated with only Kerosene as its fuel and the exhaust is studied. Then gradually Hydrogen was taken up to the combustor and the exhaust was studied. The proportions of the hydrogen injection was increased and the same studies were carried out. The observation is that, with the injection of Hydrogen the pollutant level in the exhaust reduce significantly.

The magnitude change is shown in Table 1.

TEST RUN	KEROSINE (MLPS)	HYDROGEN (MLPS)	Co (%)	HC (ppm)	NO _x (ppm)
1	16.67	0	9.87	9200	162
2	15.83	0.83	9.12	8071	156
3	15.00	1.67	8.65	6793	134
4	14.42	2.5	8.35	4643	110
OVERALL DECREASE			1.52	4557	52

Table 1:ExhaustCharacteristics

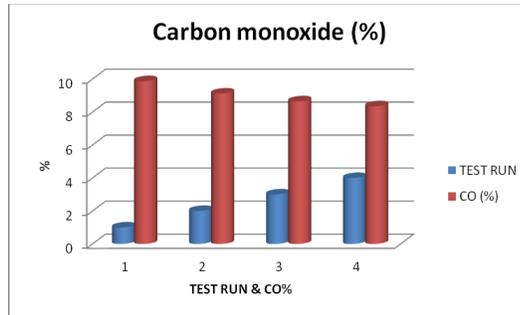


Fig7.1 Carbon monoxide reduction

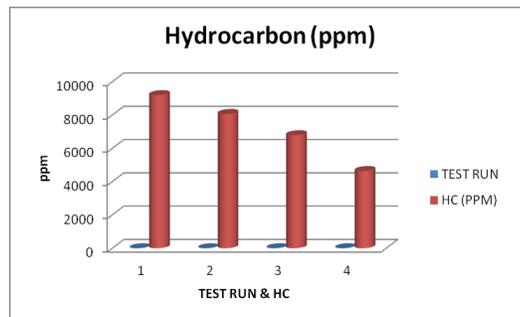


Fig7.2 Hydrocarbon reduction

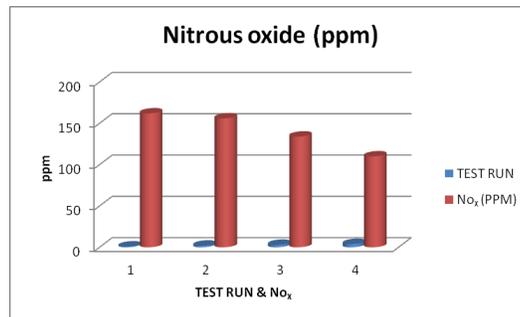


Fig7.3 Nitrous oxide reduction

As shown by the comparison charts that the amount of pollutant level in the exhaust decreases significantly with the injection of Hydrogen into the combustor. A positive result as expected.

Ozone formation

Now the nascent oxygen thus separated from water is released into the atmosphere which reacts with Oxygen molecules to produce Ozone.

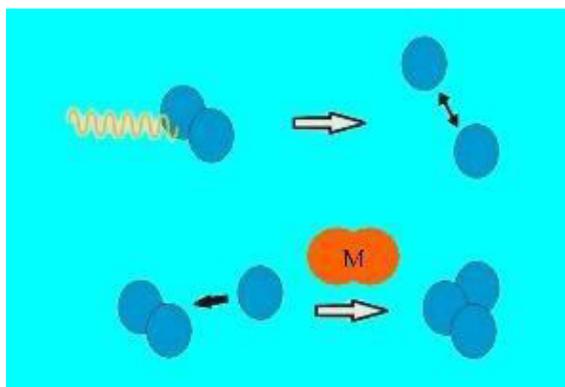


Fig7.4Ozoneformationcycle

The probability of the formation of ozone is around 30-40% in the open atmosphere. The probability is less because the nascent oxygen atom is unstable and it combines with other gas compounds.

If the same reaction is made to take place in a closed controlled environment at a temperature of about 600-800° C, the probability of formation increases to 85-95% because the combination of the nascent oxygen with the other molecules is avoided and it only reacts with oxygen alone.

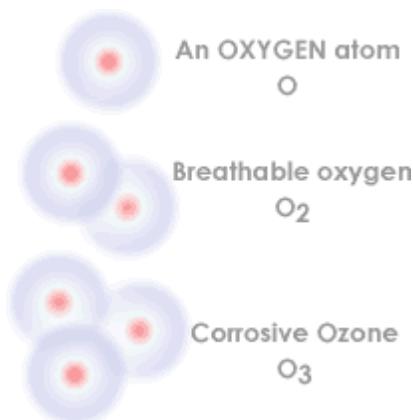


Fig7.5Oxygencompound

ADVANTAGES:

The advantages of the project are

- ✓ Pollution reduction
- ✓ Reducing hydrocarbon fuel usage

- ✓ Ozonereplenishment
- ✓ Increasedefficiencyoftheengines

Pollutionreduction

The major pollutantssuch as Nitrousoxide, Carbon monoxide and Hydrocarbon levelsfall considerably. This is a great advantage of the project; these gases are responsible for many harmful effects on the environment and also on the humankind.

Reducedhydrocarbonfuelusage

As we are going to add hydrogen as an additional fuel to the combustion chamber it reduces the amount of fuel required. The energy density of hydrogen is high thus lesser amount of fuel is sufficient for the production of the same amount of thrust.

Ozonereplenishment

The nascent oxygen atom released combines with oxygen molecule to form ozone. This replenishes the ozone layer at the same time reduces the pollution. This protects the humankind, the replenishment is done quickly otherwise at normal times it will take decades.

Increasedengineefficiency

The injection of hydrogen increases the combustion efficiency and thus increases the engine efficiency. At ground level to oxygen can be diverted into the combustion chamber which will boost up the engine efficiency at will.

REFERENCES:

1. D.DANG, M.STEINBERG – 1979 Application of the Fusion Reactor to Thermo chemical-electrochemical Hybrid Cycles and Electrolysis for Hydrogen Production from Water. V.
2. G.L. Juste – 2005 Hydrogen Injection As Additional Fuel In Gas Turbine Combustor. Evaluation Of Effects.
3. Houcheng Zhang, Guoxing Lin, Jincan Chen* - 2010 Evaluation And Calculation On The Efficiency Of A Water Electrolysis System For Hydrogen Production
4. Li Zhou Progress and problems in hydrogen storage methods
5. M.A. Deluchi – 1989 Hydrogen Vehicles: An Evaluation Of Fuel Storage, Performance, Safety, Environmental

mpacts. AndCost

6. S.Contreras–
1997HydrogenAsAviationFuel:AComparisonWithHydrocarbonFuels