

**REAL-TIME DEVELOPMENT OF EMBEDDED SYSTEM:  
TARGETTING AND CONTROL SYSTEM FOR SEMI-  
BALLISTIC VEHICLE FOR RAPID AND PRECISE SEARCH  
AND RESCUE APPLICATIONS**

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**ABSTRACT:** The Philippines is one of the most disaster-ridden nations on the western edge of the Pacific and along the circum-pacific seismic belt, or is called the Ring of Fire, which makes our nation powerless against an assortment of cataclysmic events. As a calamity-oriented country, this investigation considered the use of Unmanned Aerial Vehicles (UAV) to improve information. This paper is talks about the massive advance in the development of integrated system applications. Inserted System is a piece of equipment that incorporates hardware and product programming. It has been widely used and its plan has become a significant field of control design, notwithstanding the provision of a constant correspondence framework for observation, follow-up and verification of applications. The control framework runs on the installed equipment stage, which incorporates the microcontroller, the sensors used for route and natural checks and remote correspondence. The plan recalls the operation of the UAV for its proper operation and the establishment of secure flight controls under temperamental conditions. In addition, the project considers the best way to achieve the modification process by finding the most appropriate method of repairing wings using methods for height, length and mental edges (Heading, Pitch, and Roll) and successful paths from the target. The system is ideal for collecting, discharging and analyzing data from a few tools mounted in an unmanned aerial vehicle.

**KEYWORDS:** MEMs Vibratory Gyroscope, Wireless Sensor Networks, Embedded System, UAV

## 1.0 INTRODUCTION

The territory of the Philippines is surrounded by the Pacific Ocean, which is why the area is prone to natural disasters. Battles of disaster-prone nation, philanthropy associations in the Philippines have begun to use unmanned aeronautical vehicles (UAVs) to upgrade the information assortment and data undertakings for rapid assessment of basic zones, in particular for storm damage and hazard reduction. The innovative work of Unmanned Aerial Vehicles using the Embedded System has been inspired by directional applications. The framework created has the accompanying advantages over the traditional one, keeping an eye on the flying machine and, above all, reducing the risk to human life. UAV systems are commonly used for search and rescue operations. For non-military personnel operations, key roles were listed for fire-control, search and rescue duties, agribusiness precision, support for fire-fighting [1]. Continuous use of Article Recognition requires a high processing power, even to distinguish a solitary object, significantly less when many objective object models are included, particularly when items are included [2]. Investigation shows that the UAV system is used to use the UAV insight method to provide an ongoing communication process for identification, follow-up and analysis of applications.

The Embedded Module of the Wireless Sensor Network consists of the following areas: the Air Section and the Ground Section. The Air Section consists of the Magnetic Compass Sensor, the Micro-Electro-Mechanical Sensors (MEMs), the GPS, the Camera, the Servo Motor and the Parachute Controllers, and the latter comprises the Video Recording, Receiver and Monitoring. In addition, this system helps to control and protect maritime, land and arms forces in any region for hunting and rescue tasks.

Embedded systems used are categorized on the basis of reliability and operational specifications such as real-time and networked-based microcontroller performance.

The paper discusses the hierarchical computational system architecture to support the real-time mobile computing target domain in the context of unmanned aerial vehicles (UAVs). The overall design vision includes support for stability in the face of challenges in the operating climate of UAV surveillance. Fixed-wing Unmanned Aerial Vehicles (UAVs) are a rare category of UAVs with a surprisingly long range of points of interest. While the design of this type of UAV requires substantial planning, such as open-air trials, runways and pilots tested. Those restrictions are consistent with the design for the embedded systems for fixed-wing UAVs. Since static tests are not delegated, this paper proposes a useful way of assessing the installed framework for a proper vehicle copying the dynamic model of a fixed-wing aircraft. For this reason, an experiment is carried out between a dynamic prototype of a fixed-wing flying machine, a mobile robot style tank and a motorcycle. We prove that, as opposed to

desire in prose, a flexible robot is not the ideal choice to mimic a fixed-wing UAV. Moreover, to be sure, assuming a motion without a slip. This criterion characterizes the permitted kinematic area, which limits the imitation of a fixed-wing aircraft. On the other hand, the bike model "Bank to turn impact" is like the one seen in the fixed-wing flying machine model. Moreover, the two models are not characterized when the interpretation speed is generally zero (lower impact). In the end, we propose to use a flexible robot to test the route surface and the bike to assess the layer preparation sensor of the implanted device based on fixed-wing UAV applications. [3].

The paper proposes an arrangement capable of processing elevated images from UAVs to distinguish disappointments in estates and examines the framework running on light-measured PCs and low-power recording stages. The calculation was based on the watersheds used in the OpenCV library. The arrangement was installed on X86 (AlteraDE2i-150) and Intel Edison sheets as well as on ARM engineering (Raspberry Pi 2). Results show that the proposed system is a financially sound solution to the issue of fault identifying proof in ranches and can be increasingly incorporated into UAVs for image handling. [4].

Unmanned Aerial Vehicles (UAVs) have become a significant research field due to their enormous importance and reduced scale. A significant part of the UAVs is the capacity for self-governing flights in a number of settings and purposes, and a variety of uses have been created, from military to regular citizen fields. The system proposed in this work is a novel and re-arranged partnership between the user and the UAV for self-sufficient flight, where the important calculation is carried out on the mounted PC, which reduces the reaction time and dispenses with the need for long-distance communication with the base stations. Results are issued to the both equipment at the top of the re-create [5]

## 2.0 METHODOLOGY

Introducing the Philippines to catastrophic events could be characterized as a constant, fluctuating and serious mix that has made the nation especially aware of the collapse. It necessitates a quicker emergency and disaster response by coordinating the UAV, the liquid propelled rocket. Consistent checks were carried out on the look and rescue operations. The epic develops a UAV that can be used as a technology invention capable of conveying search and rescue activities.

Figure 1 below shows that the flight framework is primarily based on the inserted arrangement of the water rocket. It incorporates the Guidance Framework, Telemetry, Control Framework Instrument, Geographic Area Direction, Recovery Framework, and Control Framework.

The payload or the flight/correspondence contains the mind of the rocket: the primary load microcontroller used for assortment and transmission of information to the beneficiary area of the ground segment. The Inertial Measurement Unit (IMU) is an electronic gadget that is used to measure specific power, rake frequency, distance, and

use as an electronic strength control, route use and position relevant to the GPS of the UAVs. The Create System uses the IMU coupled to the GPS to provide route information for heading, pitch, roll, and separation. The Global Positioning System (GPS) is an asset that facilitates route execution, where the unwavering reliability and uprightness of GPS data checks are important, making the system reliable.

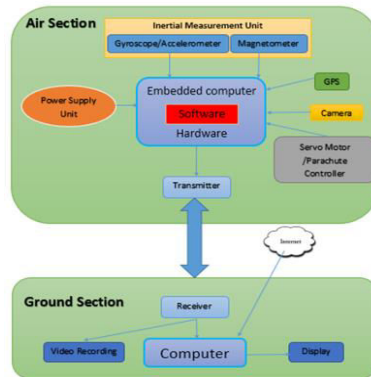


Figure 1 Unmanned Aerial Vehicle System Architecture

The camera is an electronic gadget that captures and records images. The servo motor controller is a circuit used to control the servo motor used to move the wing of the UAVs. It consists of Arduino Uno for the control of the upper wing controller, the slave controller which is dropped in the upper wing controller, the servo motor and the power supply. The parachute controller was essentially intended for the parachute configuration of the UAV. This consists of a single servo engine that can open a lock on a parachute entryway. Custom firmware is a clock that shows a trigger change to open the parachute entryway at that point. The Arduino Uno is a microcontroller board used to control and synchronize the design of the wings of the UAVs and the configuration of the parachute. The handset is a gadget which links the transmitter / beneficiary to a single package that is used for remote specialized gadgets for the air region and land section.

FPV Transmitter and Receiver Aomway DVR with dual 5.8 GB is a gadget that is actually used on a smaller scale SD card for video recording from FPV camera models. The observation section is the material segment that Matlab uses to coordinate resources and remote sensor execution.

### 3.RESULTS AND DISCUSSION

The real-time embedded system development is being presented and discussed

below.

**i. Air Section Firmware Implementation**

As shown in the outline of Figure 2 of the flow, the process for the execution of the system is visible all around the region and the ground section. The structure shows how adjustment framework is accomplished

by considering Framework design to remember the operation of the UAV for its proper operation and to render stable flight controls in a troublesome situation. The project also considers the most appropriate method of achieving the modification system by considering the most appropriate method of repairing wings by means of elevation, longitude and attitude (Heading, Pitch, and Roll) methods and by means of successful pathways from the objective.

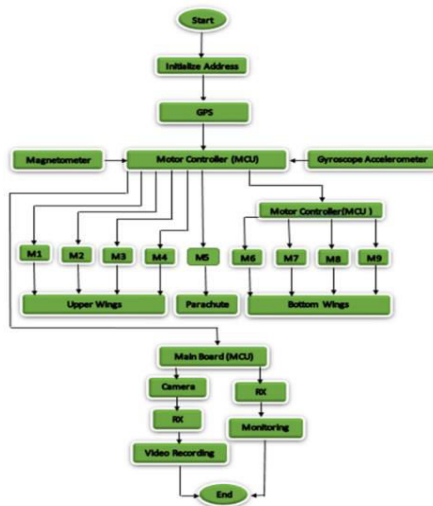


Figure 2 Firmware Flowchart

**ii. Ground Section Firmware Implementation**

As appeared in Fig. 3 stream outline, it shows the program execution of the shot tracker shows. The guide see provides all-encompassing perspectives for the development of the UAV. The shot outlook displays the shot movement as it moves while the data lumberjack records the altitude, longitude and frame of the mind edges (Heading, Pitch, and Roll) and the good paths from the objects.

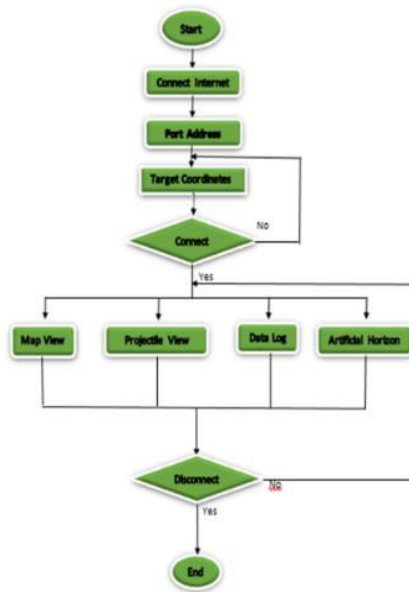


Figure 3 Projectile Monitoring Flowchart

**iii. Water Rocket Design**

Figure 4 below shows the three-dimensional arrangement of the UAV made of composite material that fuses the body of the reinforced structure, the wings, the nosecone, the weight vessel, the aeronautical structure and the support.

The airship used for the semi-ballistic UAV is a liquid rocket filled with carbon fiber tangle composite materials. The reinforced structures were nuts.

Equipment or a water tank involves an impeccable jug and a fixed wing that protects the rocket from turning and enough drag that the rocket can control itself, while the payload or the flight / interchange contains the cerebrum or the gag of the rocket. Figure 1 below shows a stage that can be divided into two parts: the air zone and the

ground section.

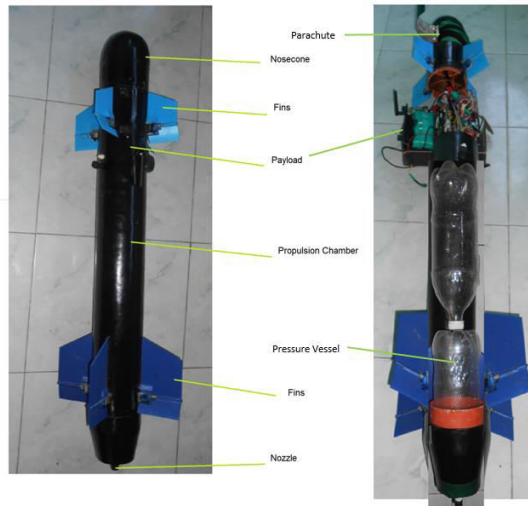


Figure 4 Implementation of UAV Air Section

Figure 4 shows the actual structure of the UAV. This requires a payload or a flight / trade comprising the psychology of a rocket: a microcontroller bearing the vital responsibility of the social event and transmitting data to the receiver territory of the planet, a camera for video recording, a battery, a GPS, an accelerometer and a compass, and a servo controller. The gear or the water tank includes the ideal owner and the correct wings that protect the rocket from turning and decent drag that the rocket may control, while the payload or the flight / exchange comprises the brain or the rocket stifler.

Mobile balances use a pole that links the blade to one side and the servo rotor to the opposite end. Four blades of a similar size for the back and a smaller size for the above braces, form and weight will be attached to the corresponding sides. The balances are made of a 3D print of conventional fiber due to its solidity, lightweight, waterproof and easy to insert. The small-scale servo motor pole is attached to the leading edge of the balances.

#### **iv. Ground Section Circuitry Implementation**

Circuit design appeared in Figure 5 shows the connections which incorporate the accompanying parts such as the Receiver, Converter, Personal Computer, and the Video

Recorder.

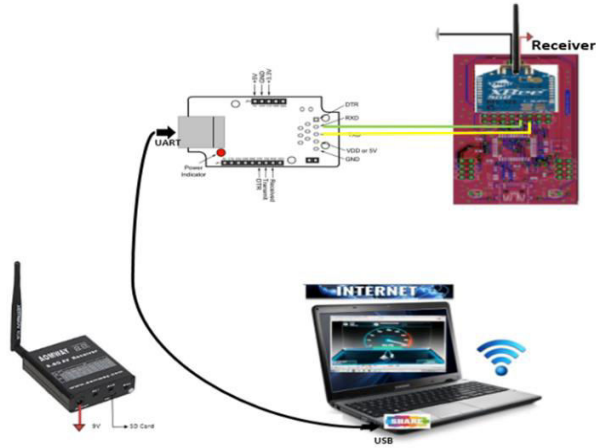


Figure 5 Ground Section Circuitry

### V. Air Section Circuitry Implementation

The circuit model shown in Figure 6 indicates the relation which includes the corresponding components such as the Embedded Computer, GPS, Camera, Transmitter, Sensors and Controller.



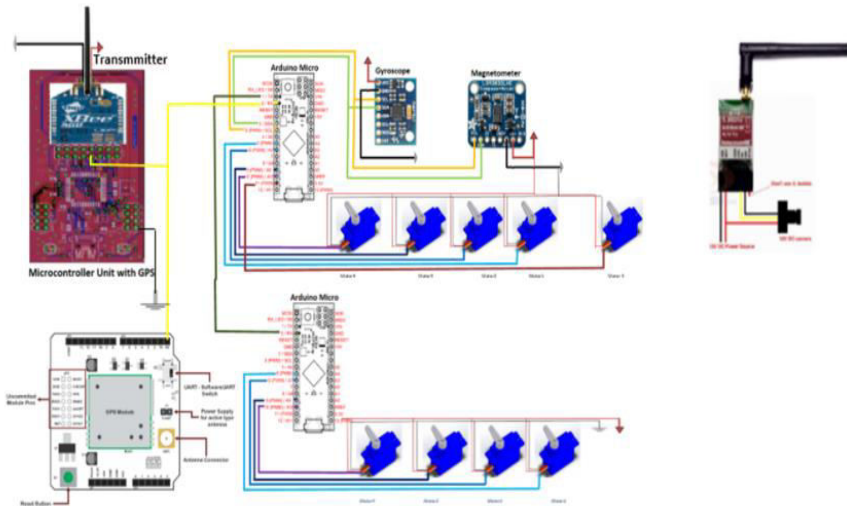
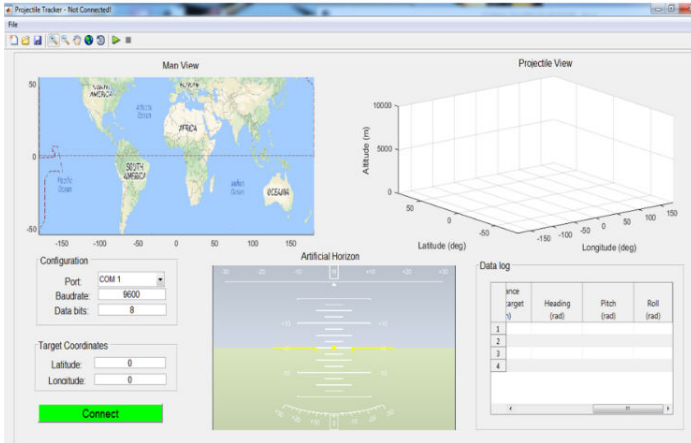


Figure 6 Circuitry of the Air Section

**v. Projectile Tracker Display Output**

Since this test is based on the Matlab 2016 framework for follow-up and analysis of shots. Figure 7 below shows a recreated shot after a presentation using it. To have the option to connect a shot tracker display and be implemented, the web button must remain green, which means that it is not yet linked.



Figure

Figure 7 Projectile Tracker Display (OFF)

Sunderneath shows the genuine outcome during the starting of the water rocket.

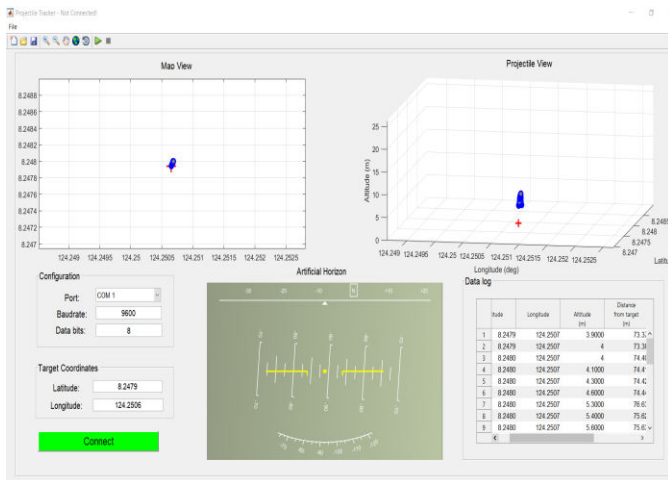


Figure 8 Projectile Tracker Display (ON)

**4. CONCLUSION**

The conceptualization and implementation of the Go-to-Target Advancement and Control System for Semi-Ballistic Vehicles for Rapid and Accurate Pursuit and Rescue Activities has been established. The integration of a multi-sensor and a blend of communications between the two segments were successfully developed in the general plan and the development of the execution of a semi-ballistic unmanned aeronautical vehicle.

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